



10th EAWE PhD Seminar on Wind Energy in Europe  
October 28-31, 2014 in Orléans, France

# Testing airfoils for wind turbines

## The 3D challenges



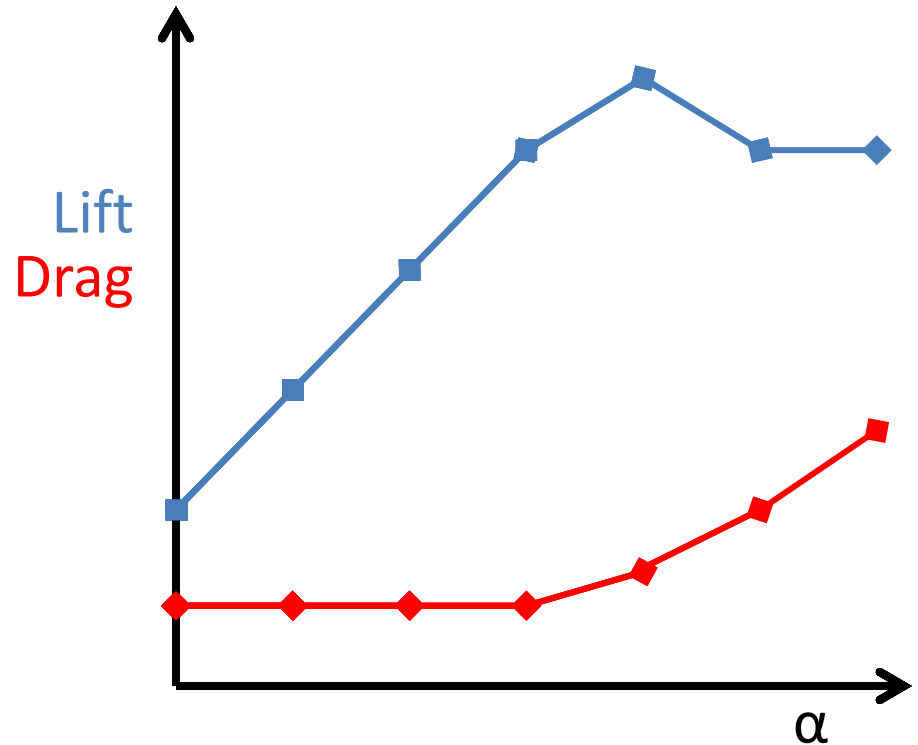
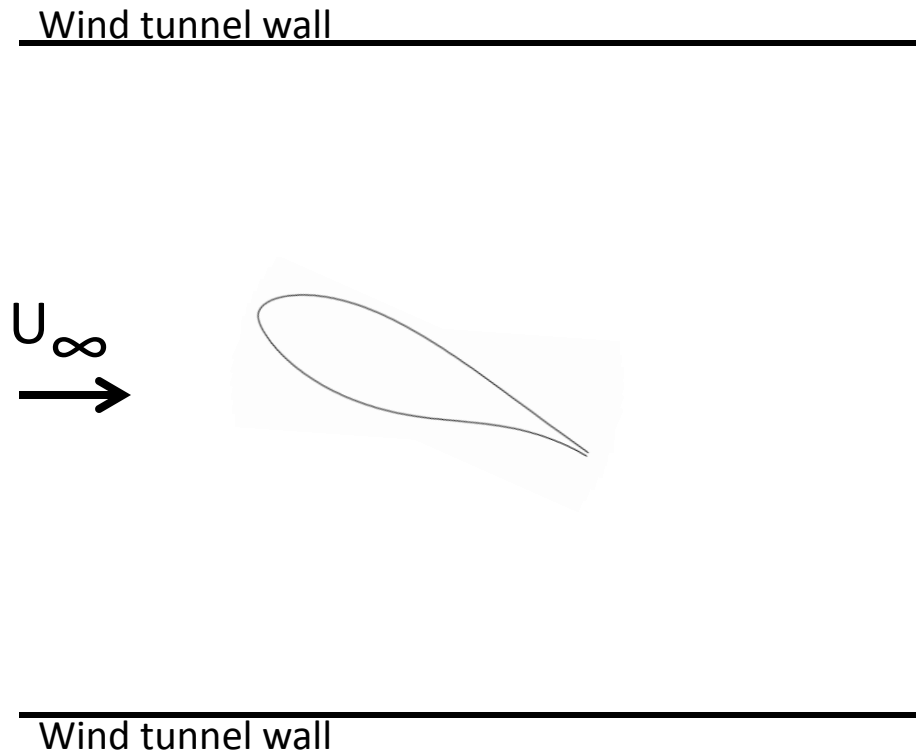
**Marinos Manolesos**

National Technical University of Athens  
Fluids Section

# Introduction



## 2D wind tunnel polars for airfoils

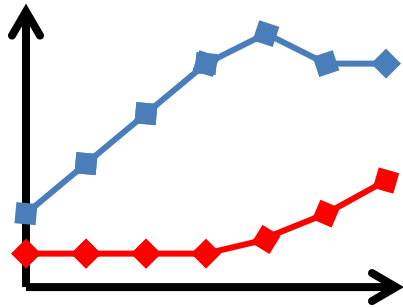


Incompressible, “external”, wall bounded flow,  $Re \sim 10^6$

# Introduction



## 2D wind tunnel polars for airfoils



Blade Element  
Momentum (BEM)  
Code



- Annual Energy Production
- Loads on blades

Main assumption:

**The flow around the airfoil is 2D.**

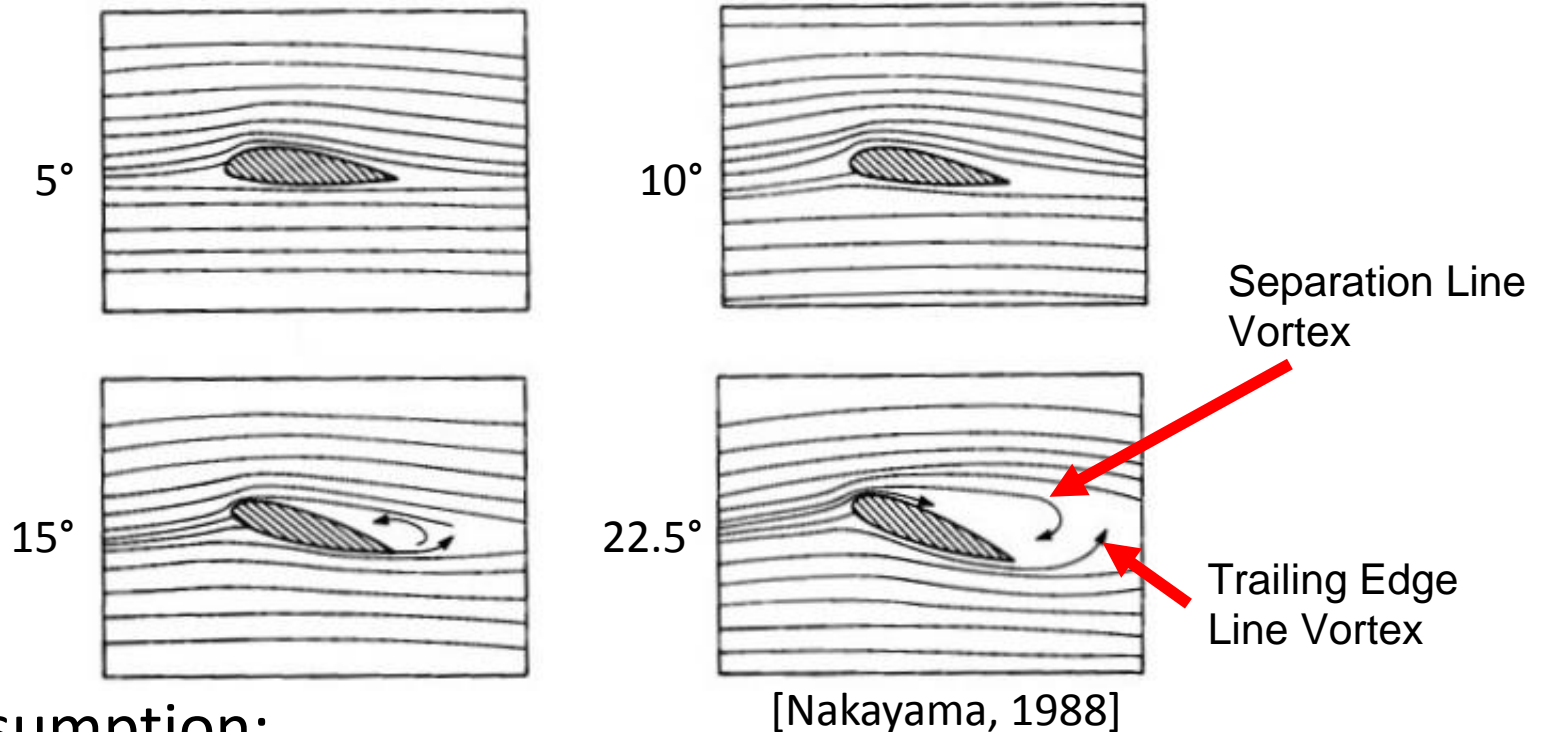
Question:

**Does a symmetric 2D set-up result in symmetric 2D flow?**

# Introduction



## 2D wind tunnel polars for airfoils



Main assumption:

**The flow around the airfoil is 2D.**

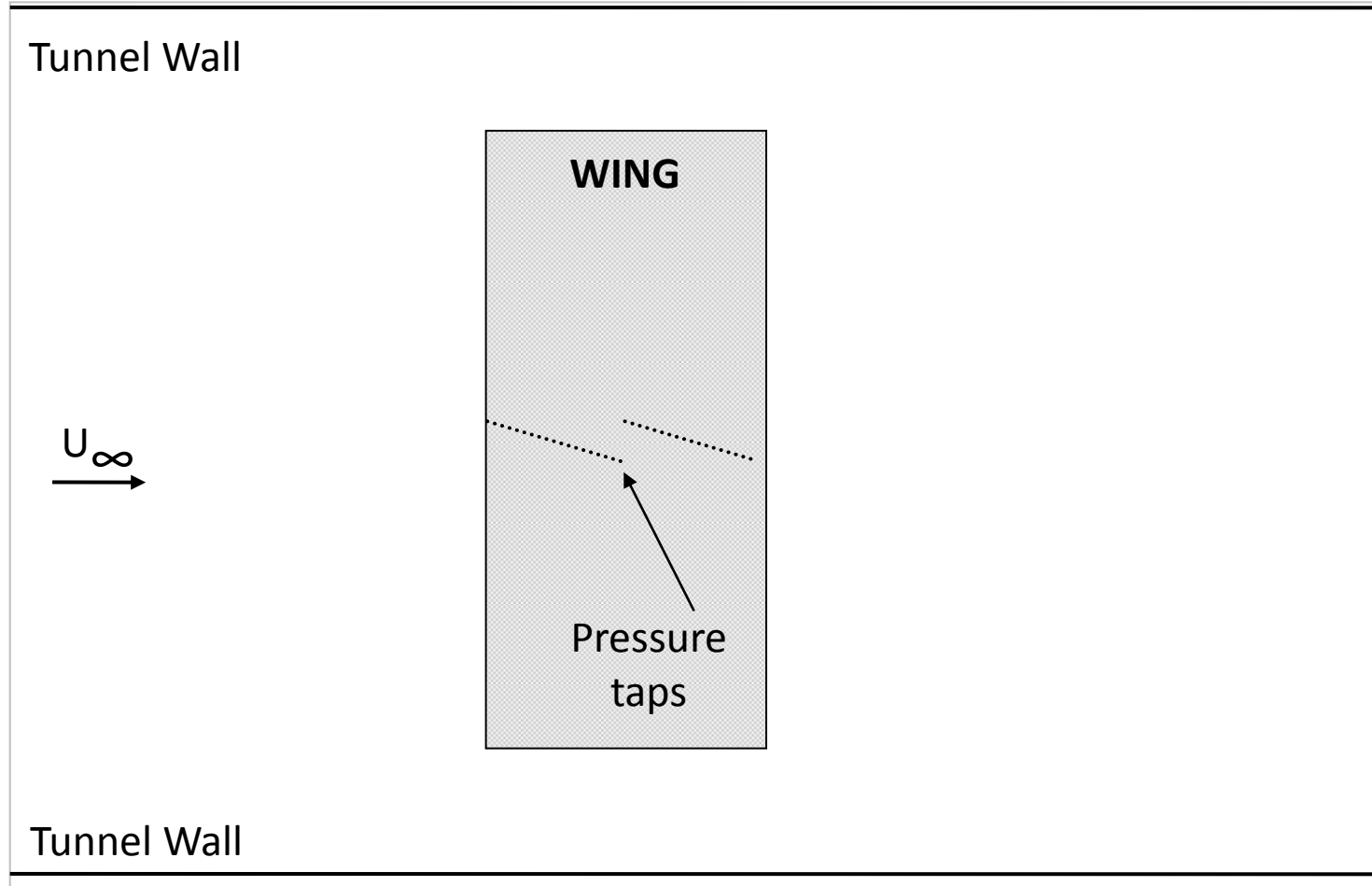
Question:

**Does a symmetric 2D set-up result in symmetric 2D flow?**

# Experimental Set Up



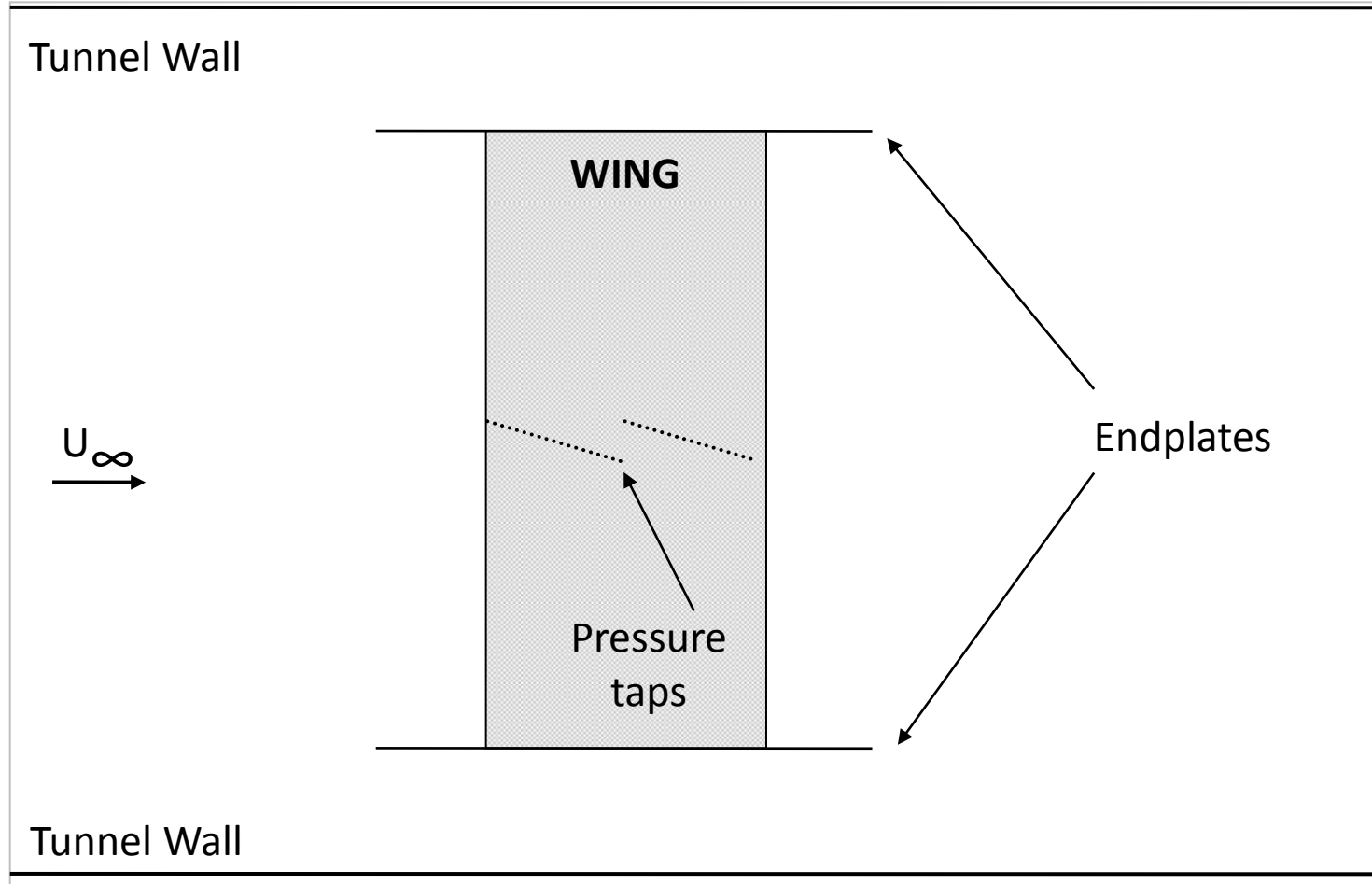
## Wing with free tips



# Experimental Set Up



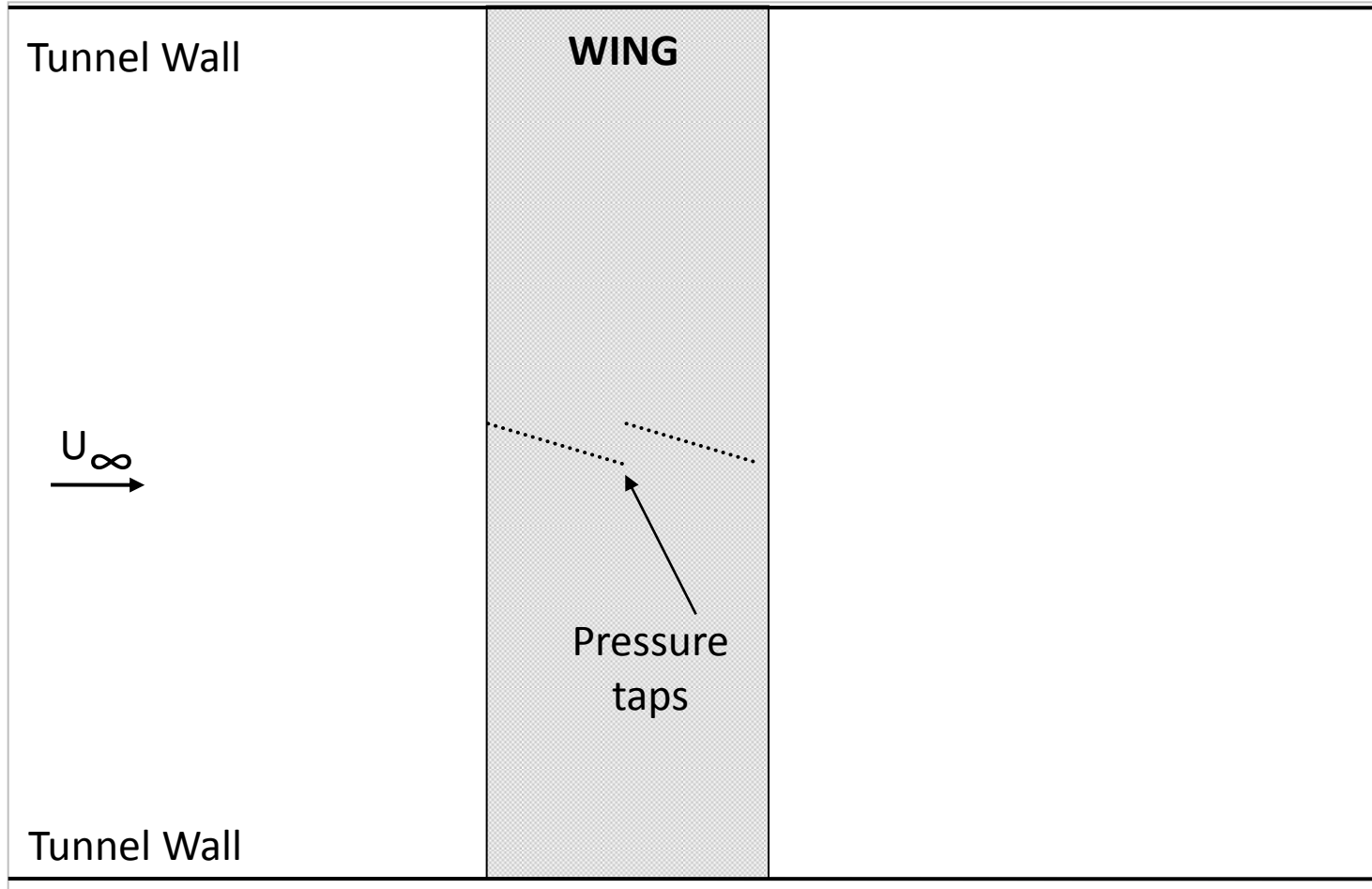
## Wing with endplates



# Experimental Set Up



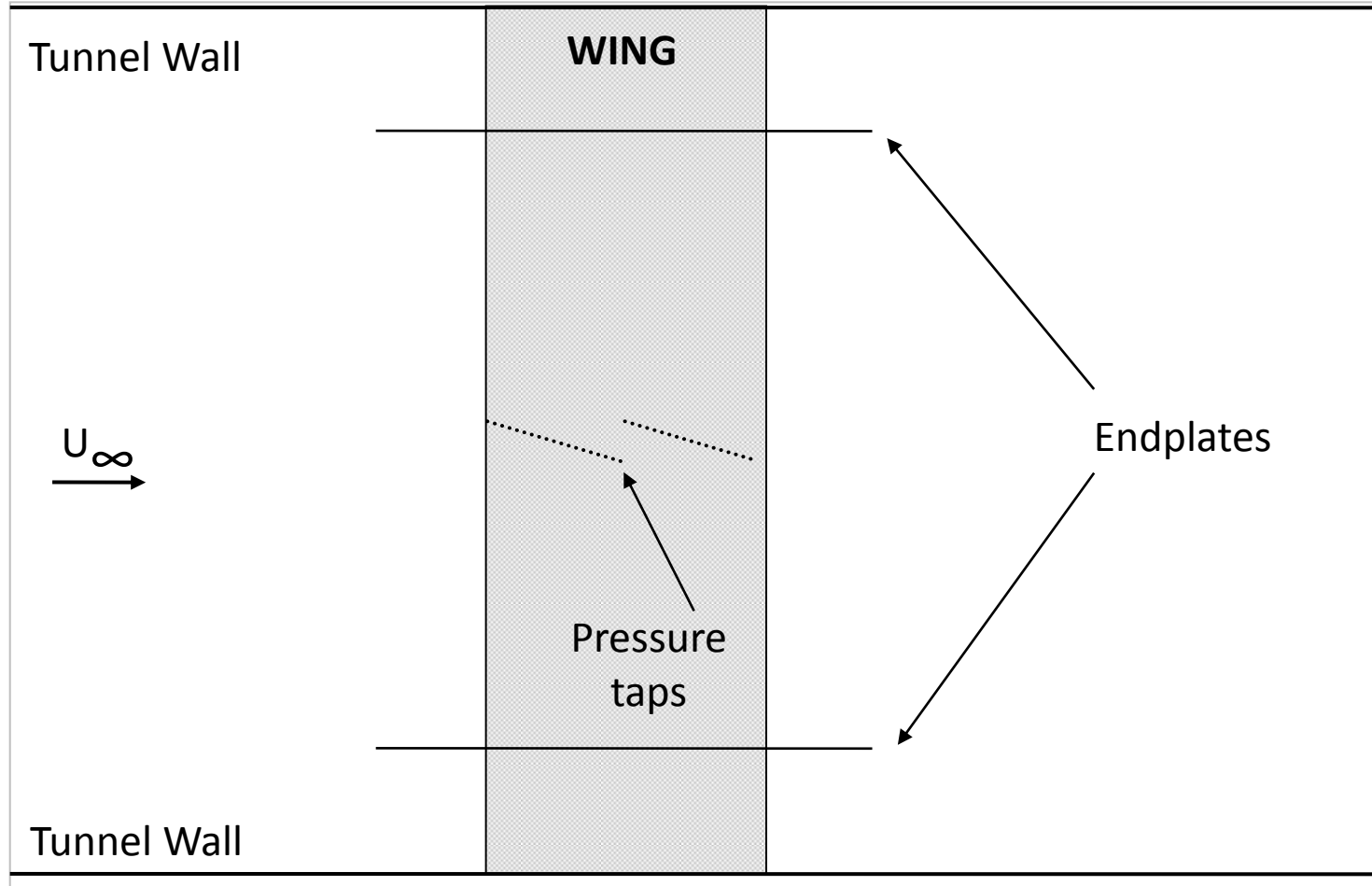
## Wall to wall wing model



# Experimental Set Up



## Wall to wall wing model + endplates



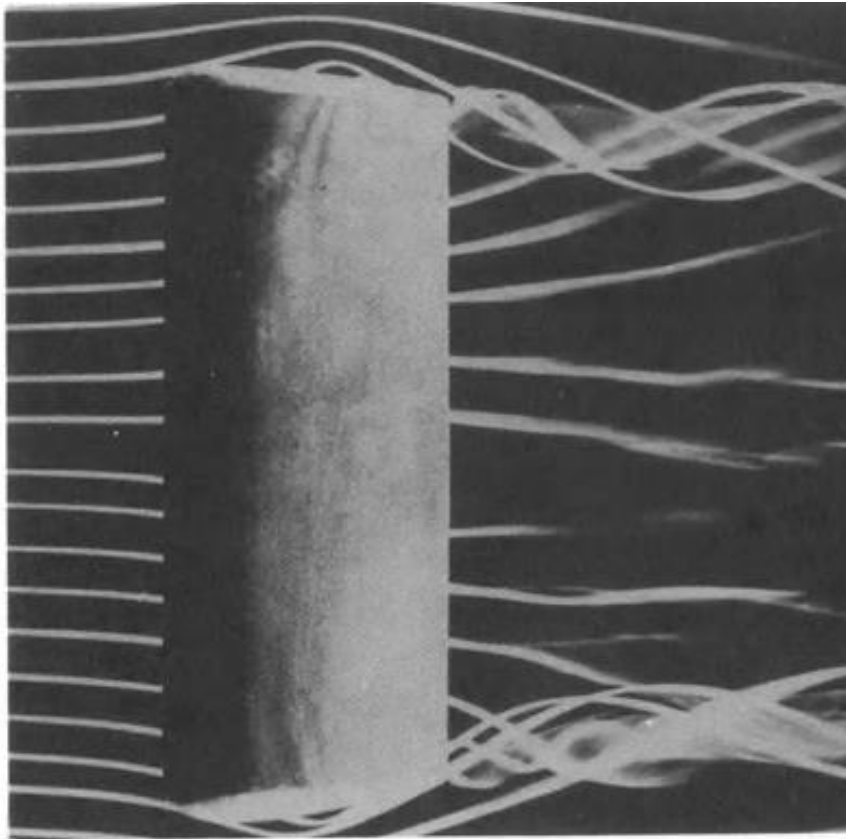


# Experimental Set Up



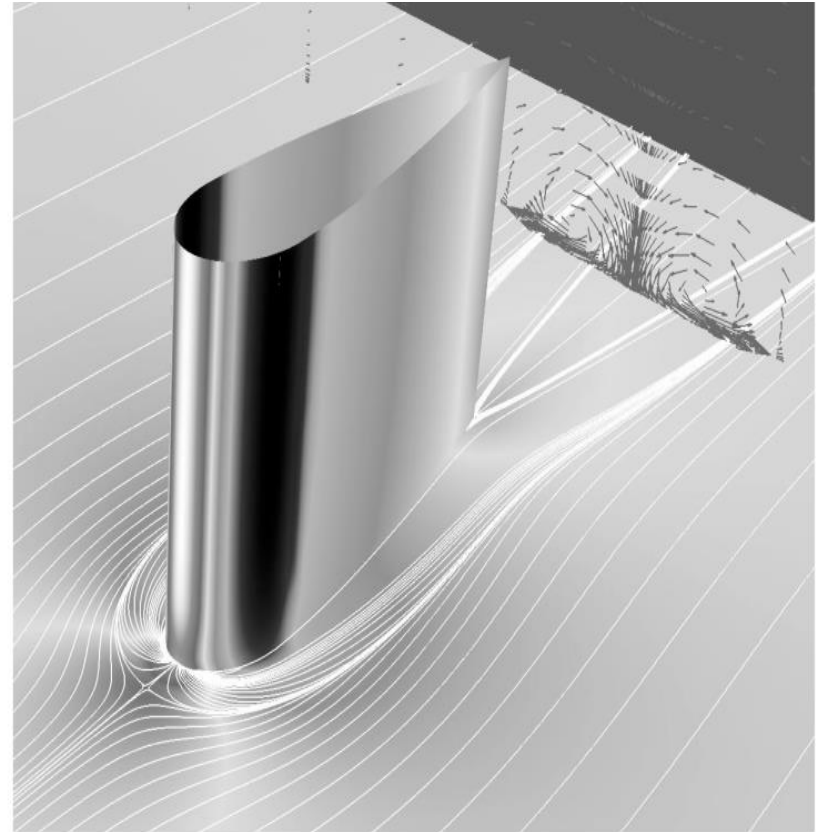
The flow is always 3D at the tips

**Free tips**



[Head, 1981]

**Wall to wall/endplate models**



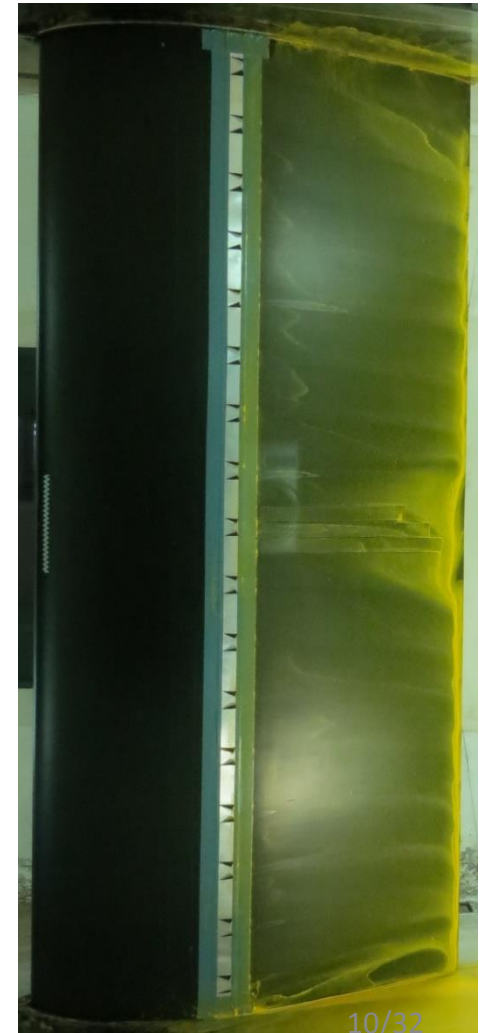
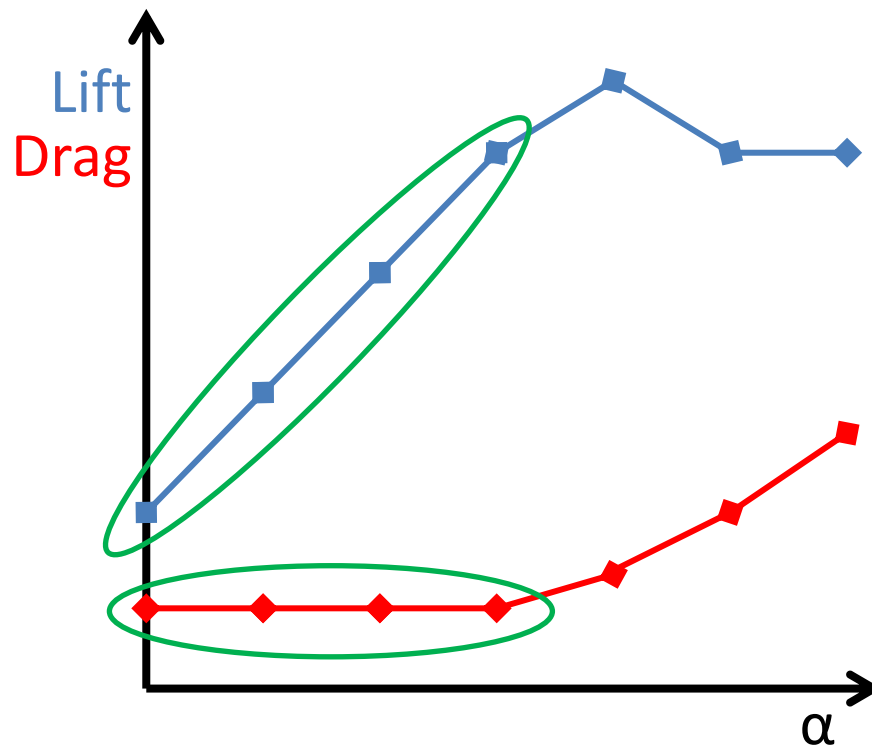
[Apsley, 2001]

# Experimental Set Up



What happens away from the tips?

2D flow under attached flow conditions



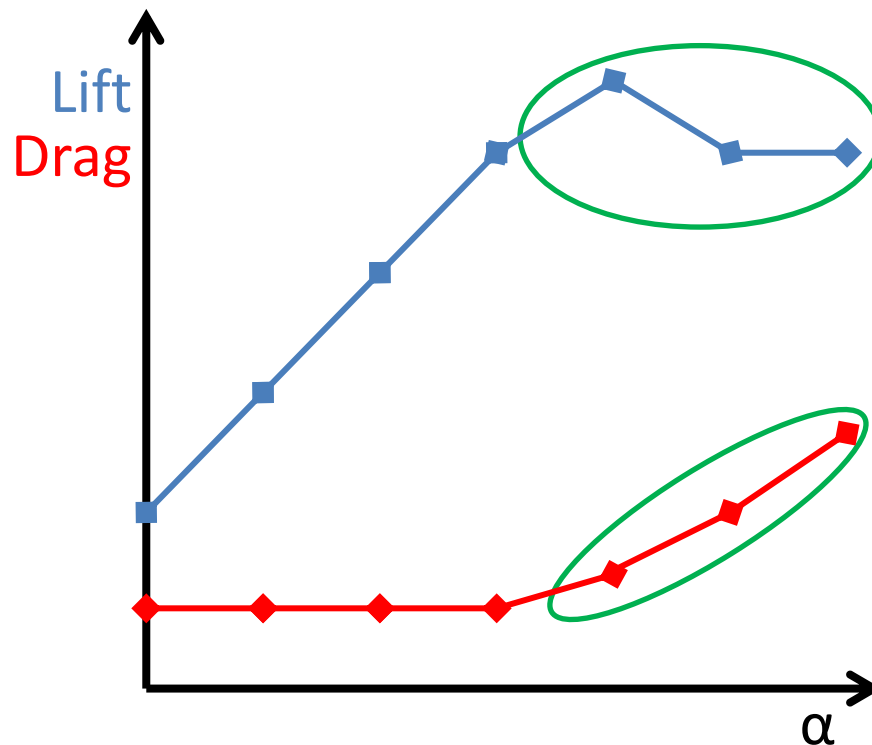
# Experimental Set Up



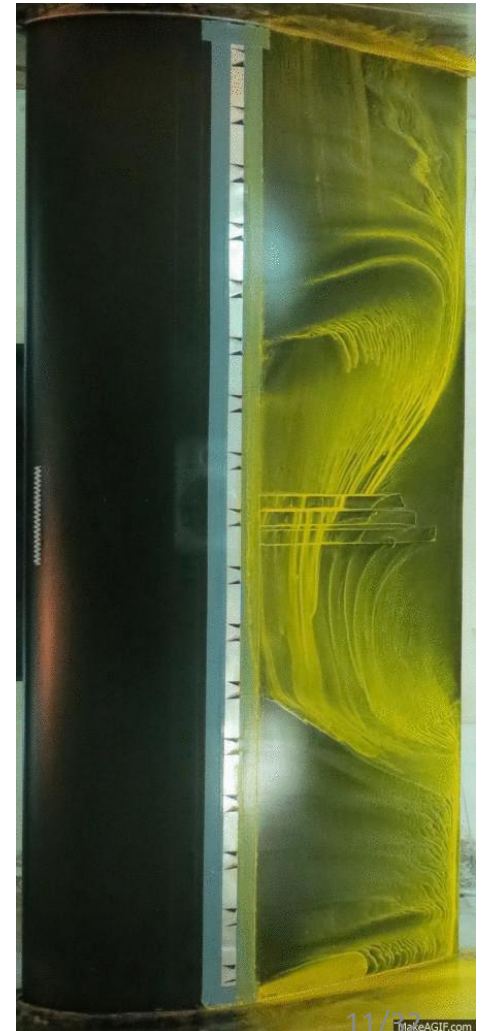
What happens away from the tips?

3D flow around maximum Lift

**Stall Cells** are formed



$U_\infty$



# Stall Cells



## Stall Cells are coherent structures of separated flow

- large scale
- consist of a pair of counter rotating vortices
- are unstable
- have been observed for a very wide range of Reynolds (Re) numbers
- are not a tip effect

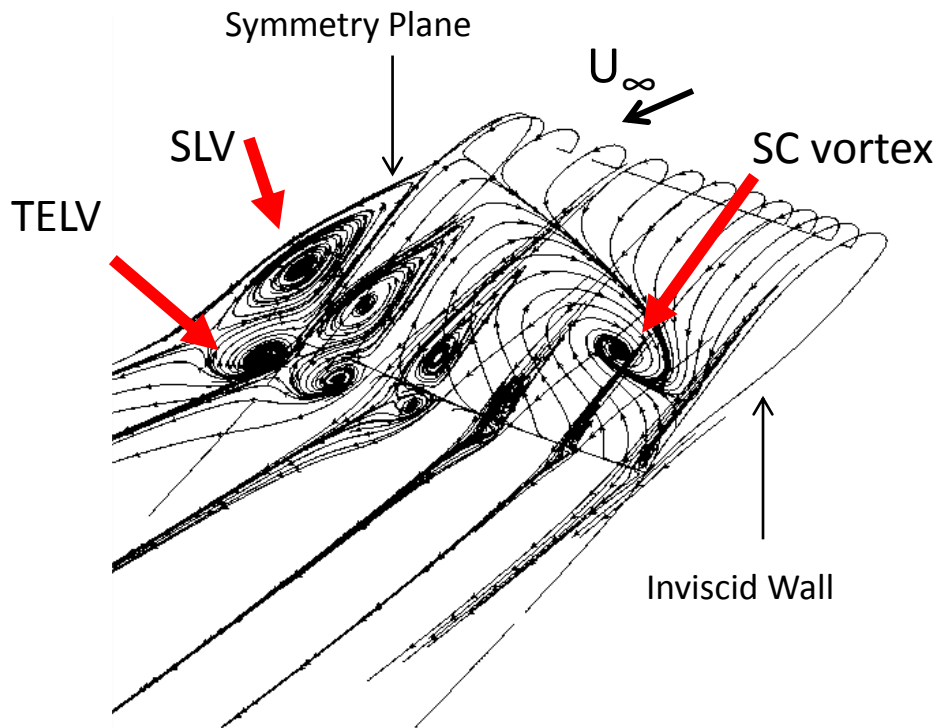
$U_{\infty}$



# Stall Cells



## Structure



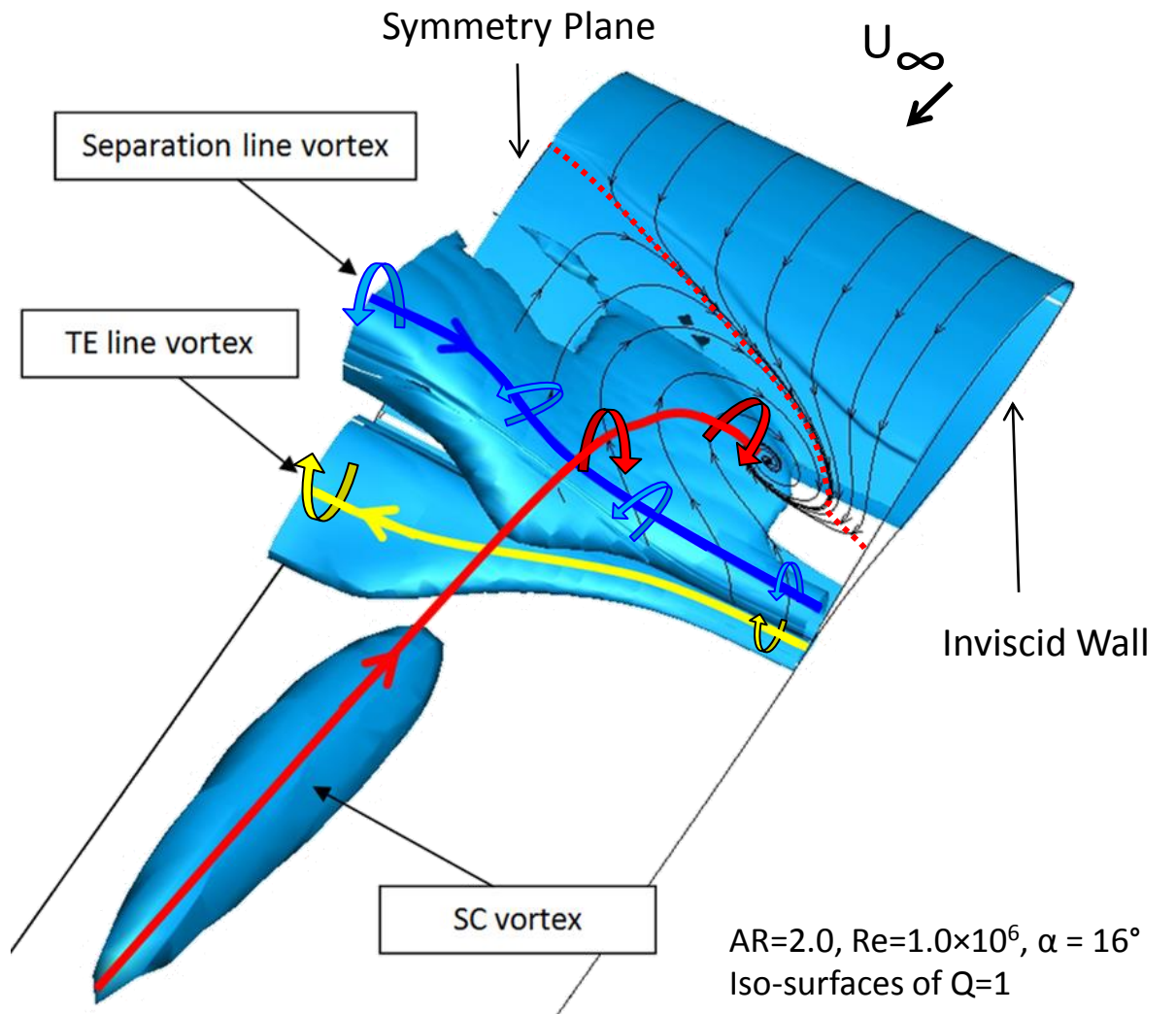
AR=2.0,  $Re=1.0 \times 10^6$ ,  $\alpha = 16^\circ$

- The SC vortex
  - starts normal to the wing surface
- The Separation Line Vortex (SLV)
  - parallel to the wing TE
- The TE line vortex (TELV)
  - parallel to the SLV but with vorticity of opposite sign

# Stall Cells



## Structure

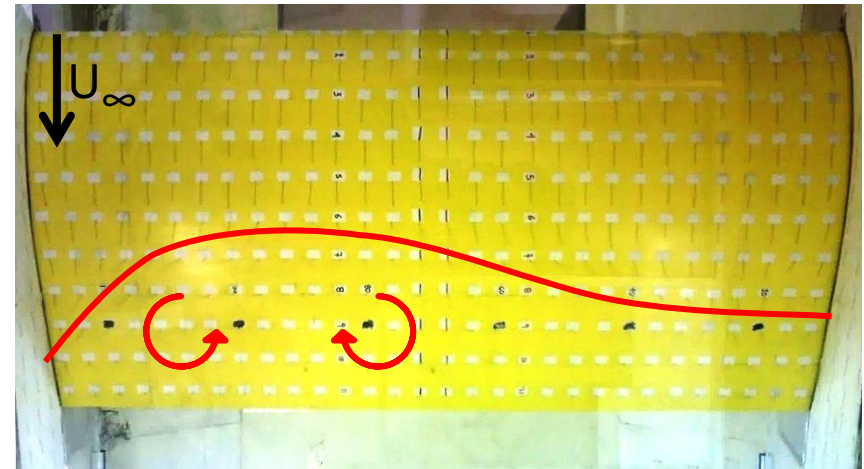
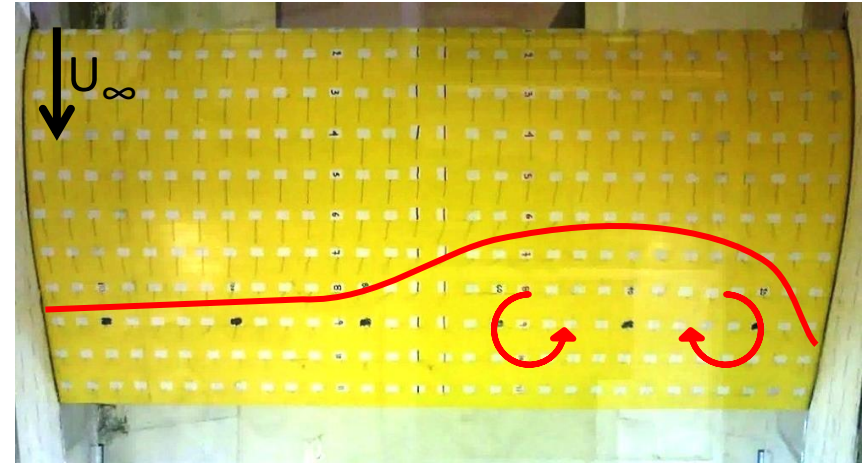


# Stall Cells



## Instability

- Their number or position can change arbitrarily



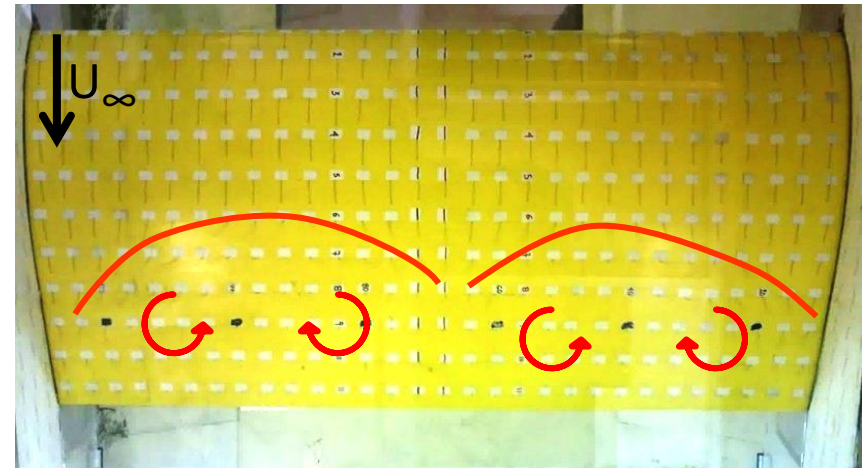
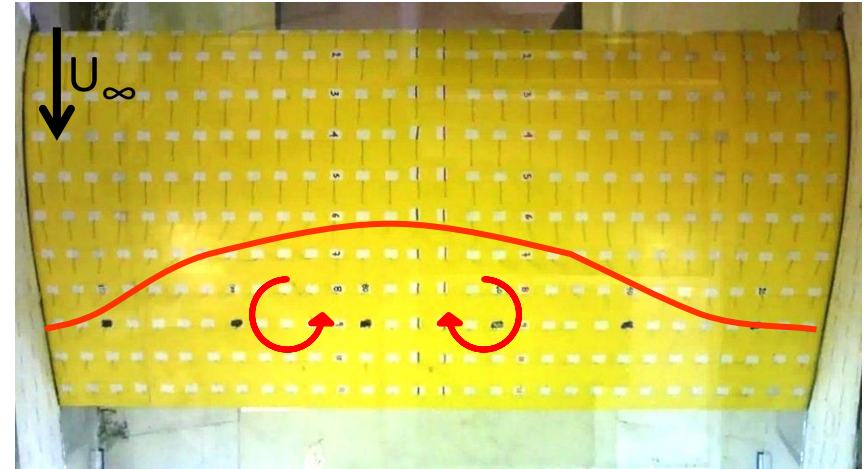
AR 2.0,  $\alpha = 11^\circ$ ,  $Re = 1.0 \times 10^6$

# Stall Cells



## Instability

- Their number or position can change arbitrarily
- Their movement/formation shows no apparent periodicity
  - No correlation with Re number, aspect ratio or angle of attack has been found.



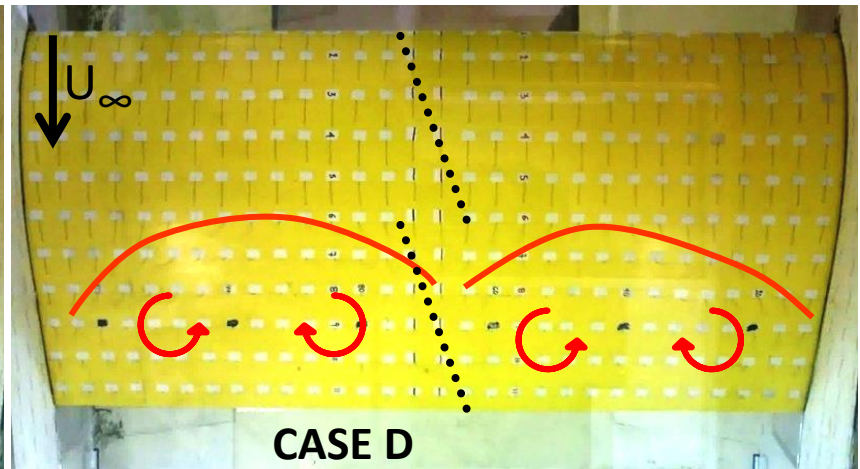
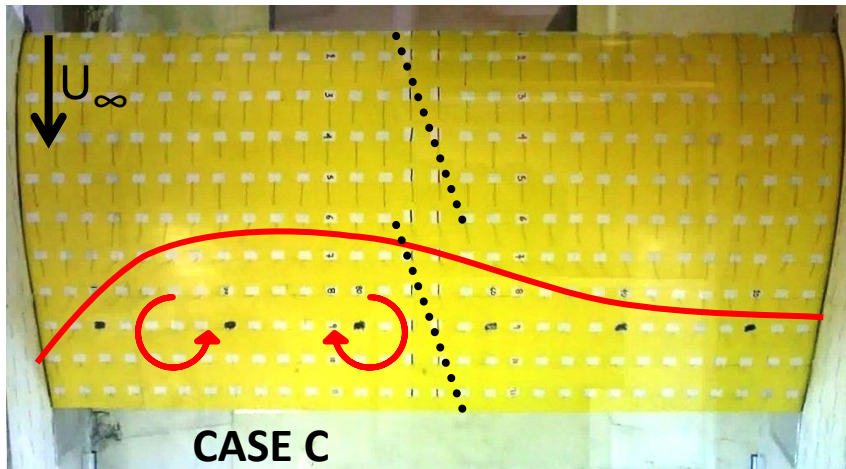
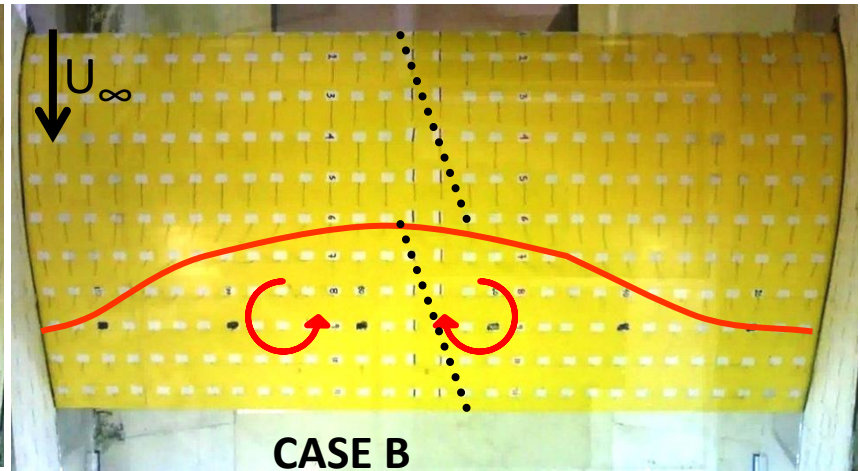
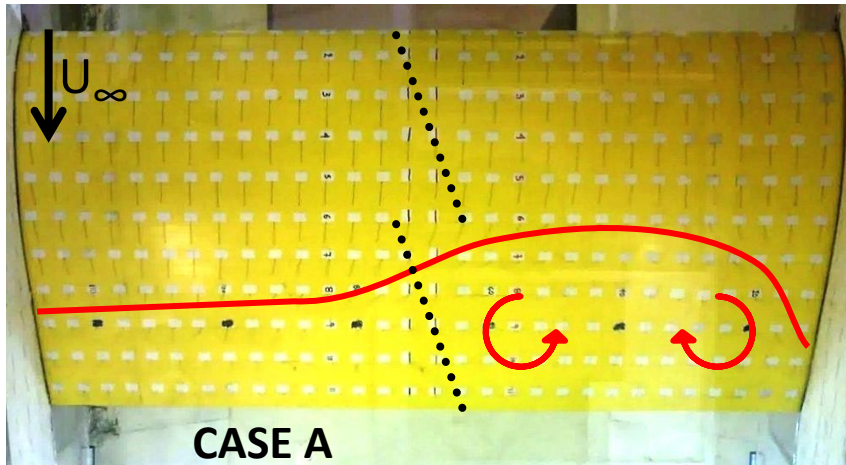
AR 2.0,  $\alpha = 11^\circ$ ,  $Re = 1.0 \times 10^6$



# Implications



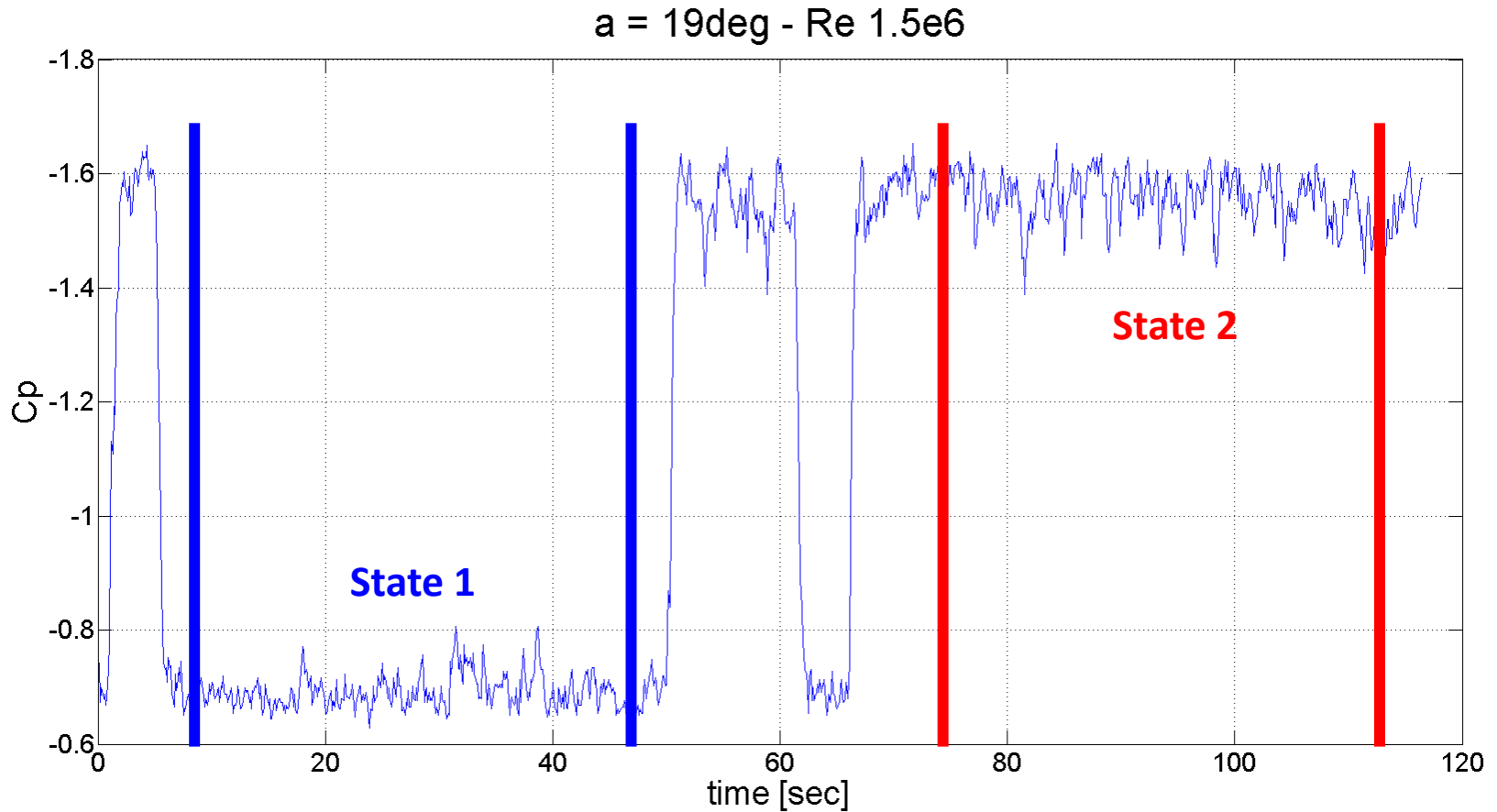
## Know what you measure



# Implications



## Know what you measure

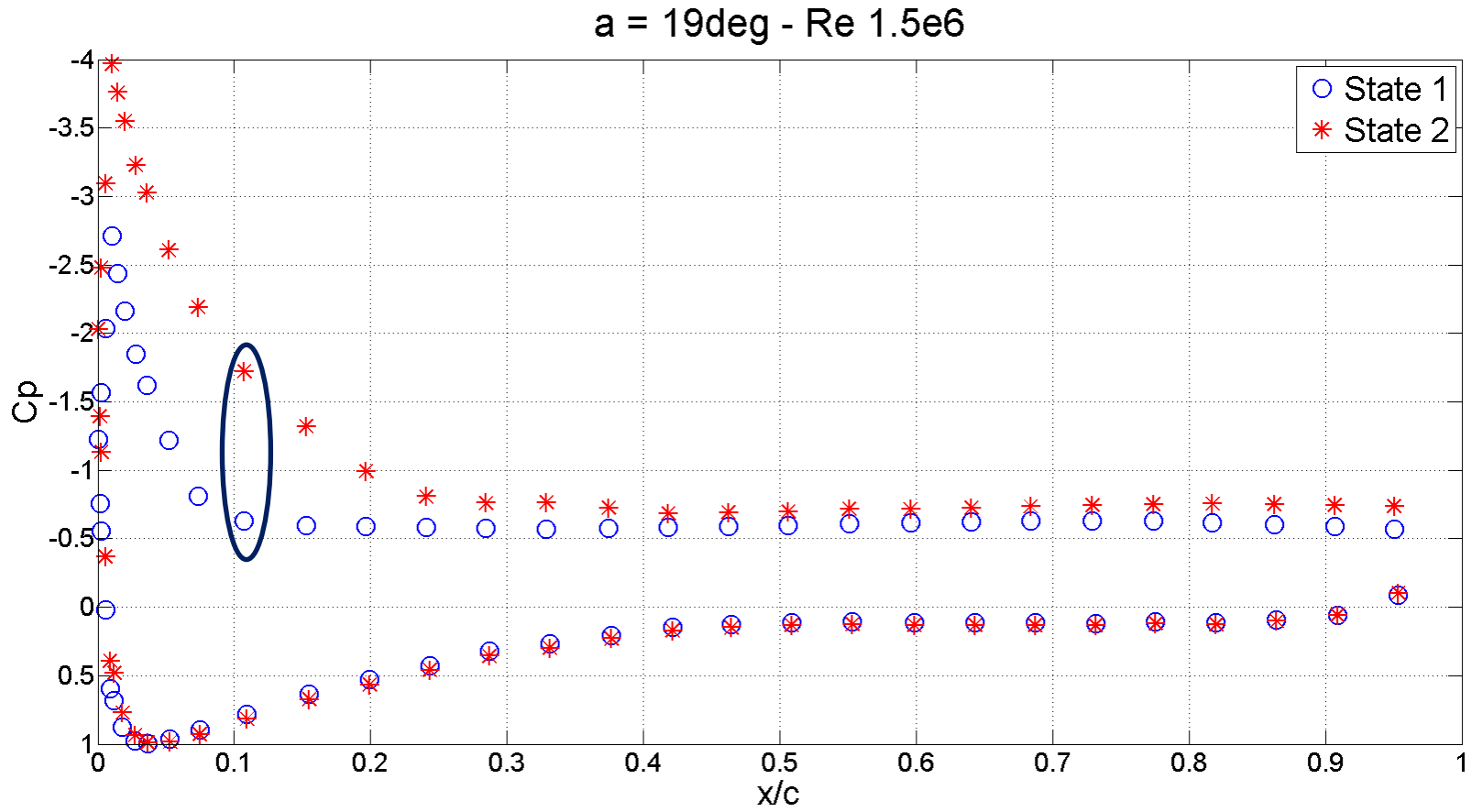


Pressure time series from tap at  $x/c = 0.11$

# Implications



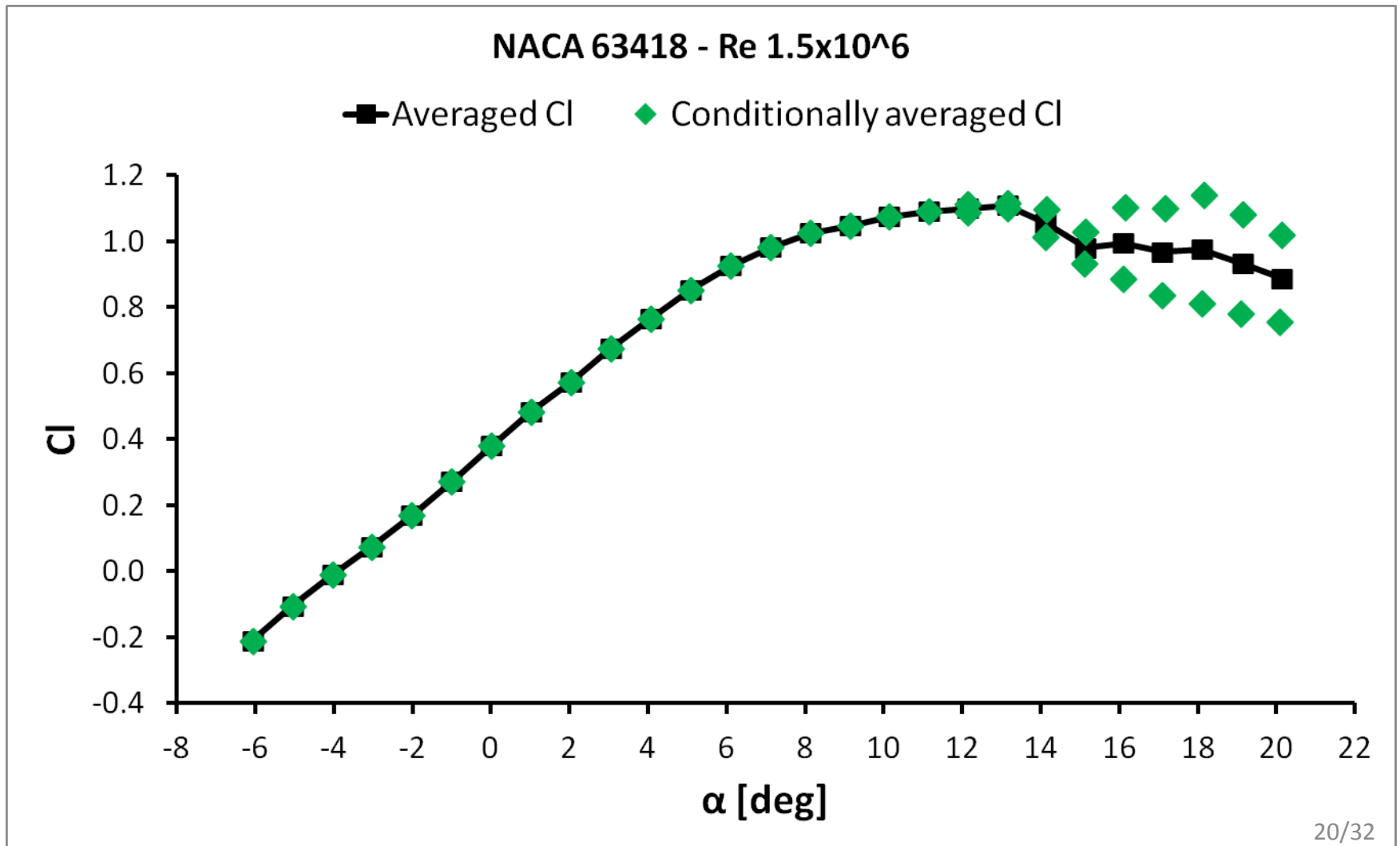
## Know what you measure



# Implications



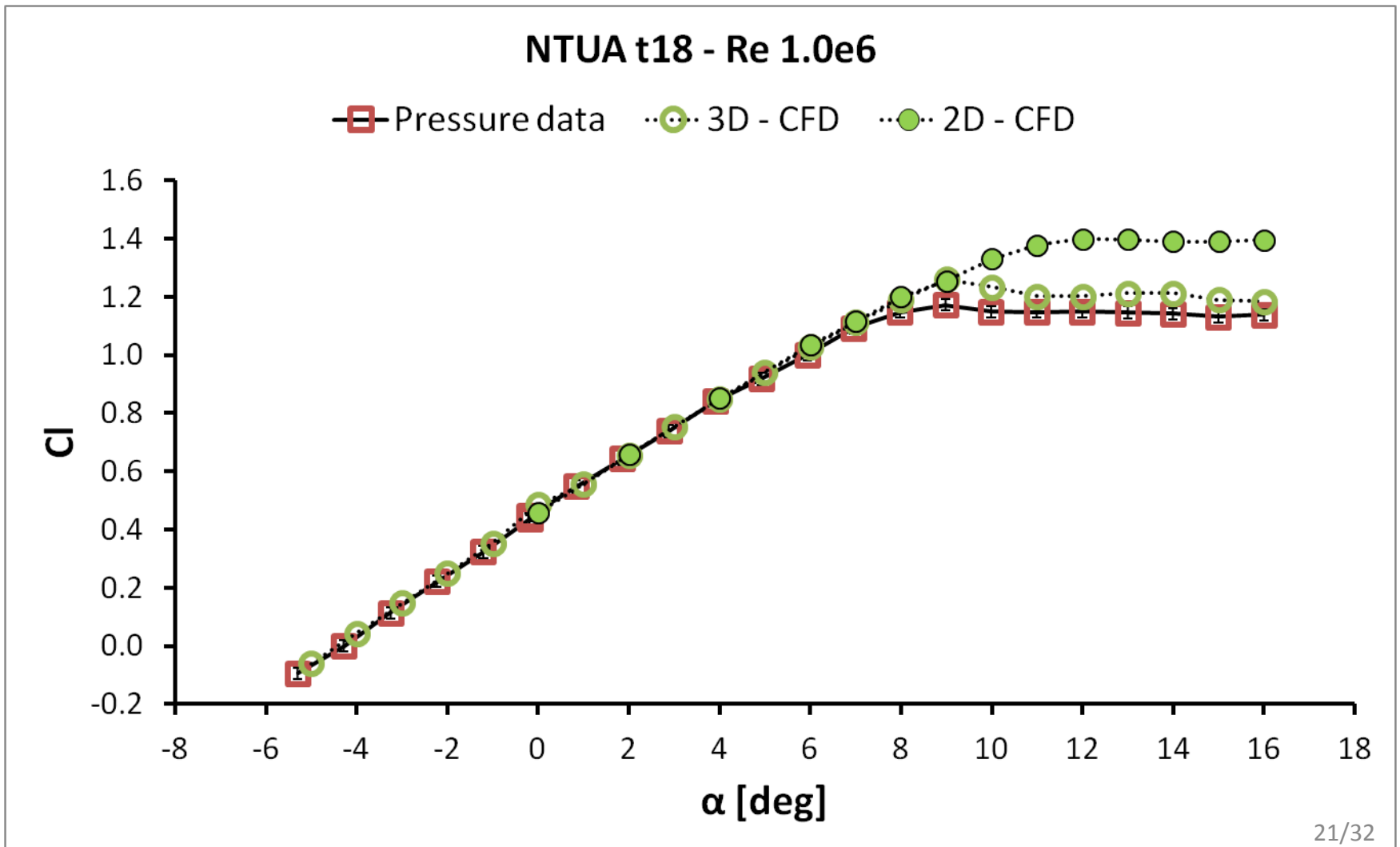
## Know what you measure



# Implications



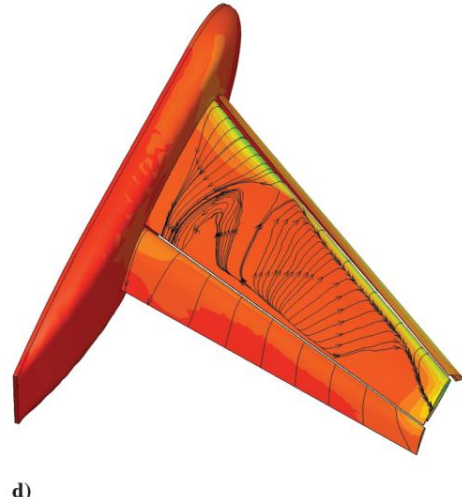
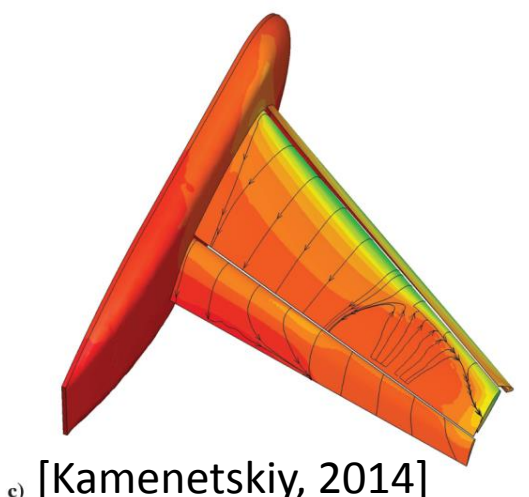
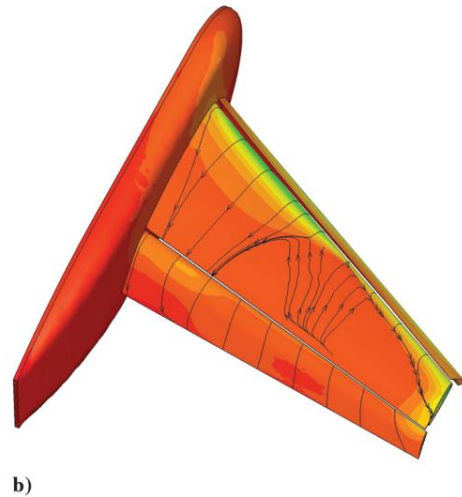
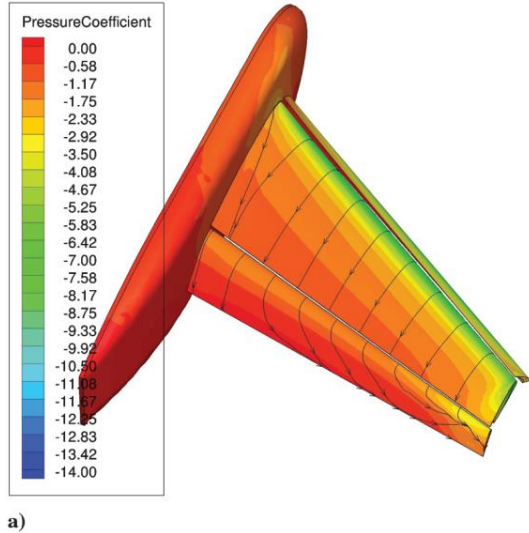
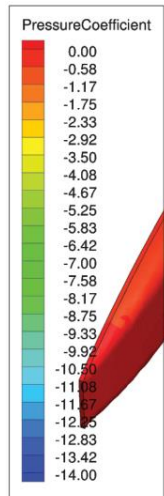
## Know what you compare to



# Implications



## Know what you compute



Multiple solutions exist for the discretized RANS equations

- Grid
- Turbulence model
- Convergence
- Initial conditions
- Implementation of boundary conditions
- Perturbations

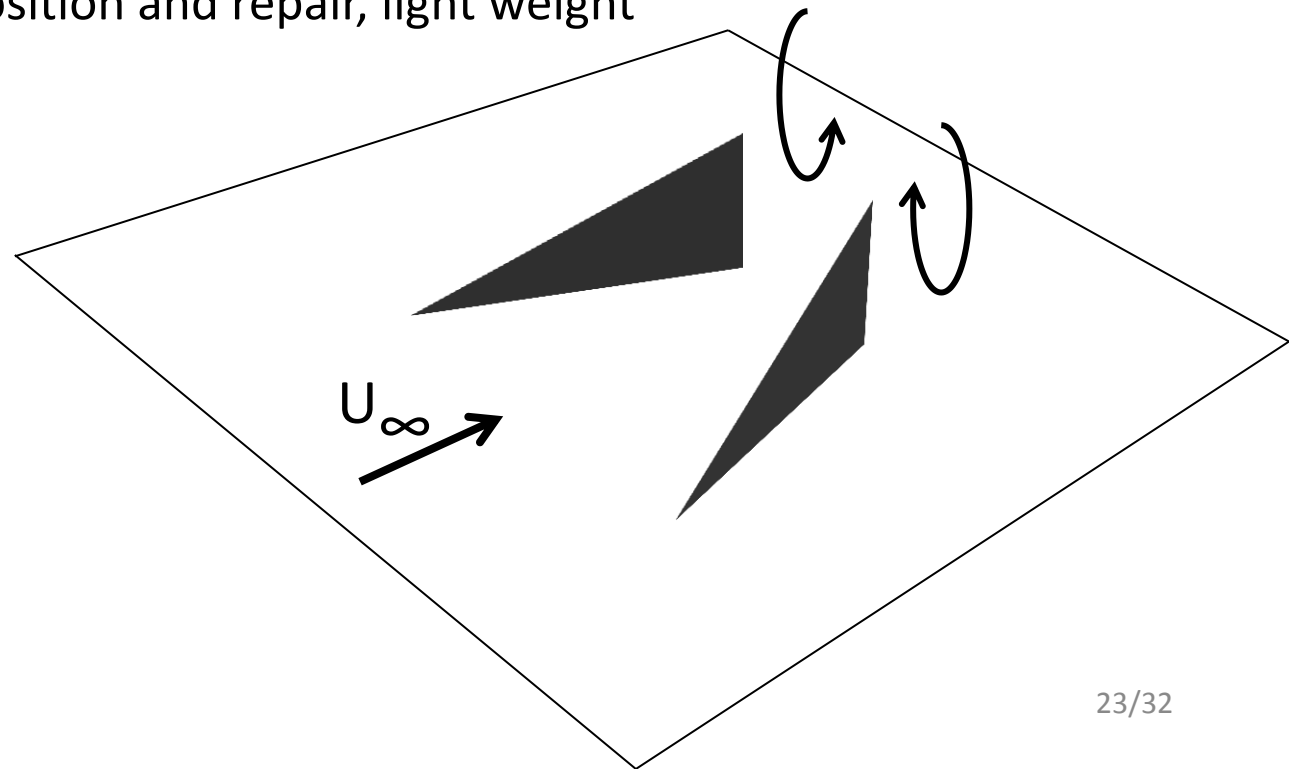
Which solutions correspond to real flow?

# Stall Cell Control



## Passive Vortex Generators (VGs)

- VGs are vanes located normal to the wing surface
- Their height is at the order of the Boundary Layer height
- They create streamwise vortices that energize the boundary layer and thus delay separation
- Easy to construct, position and repair, light weight



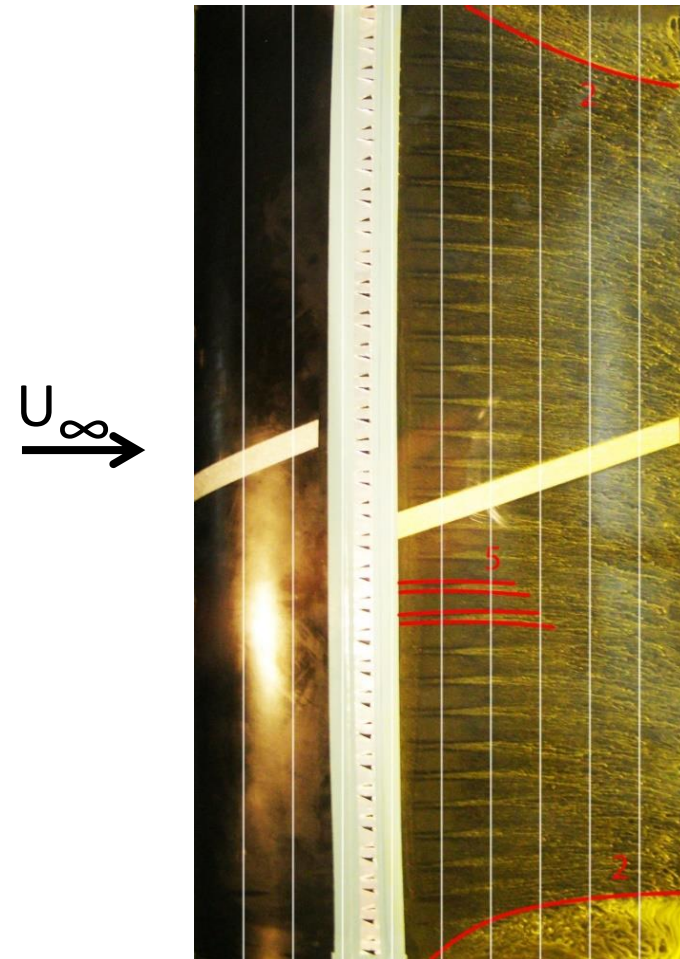
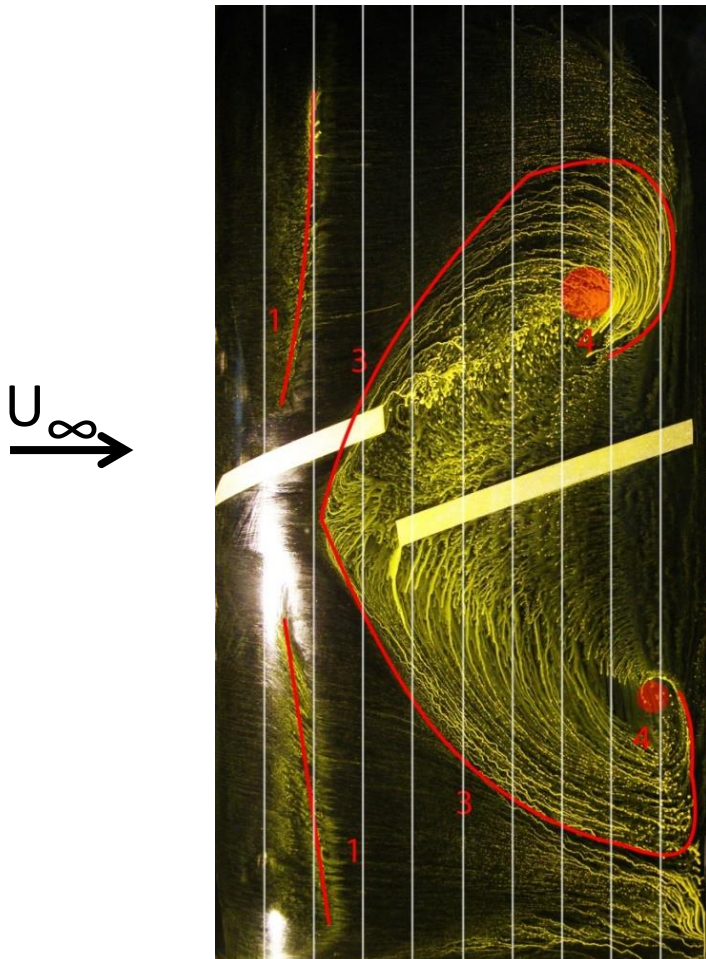
# Stall Cell Control



## Passive Vortex Generators can delay SC formation

without VGs

with VGs



$Re = 1.5 \times 10^6$ ,  $\alpha = 14^\circ$

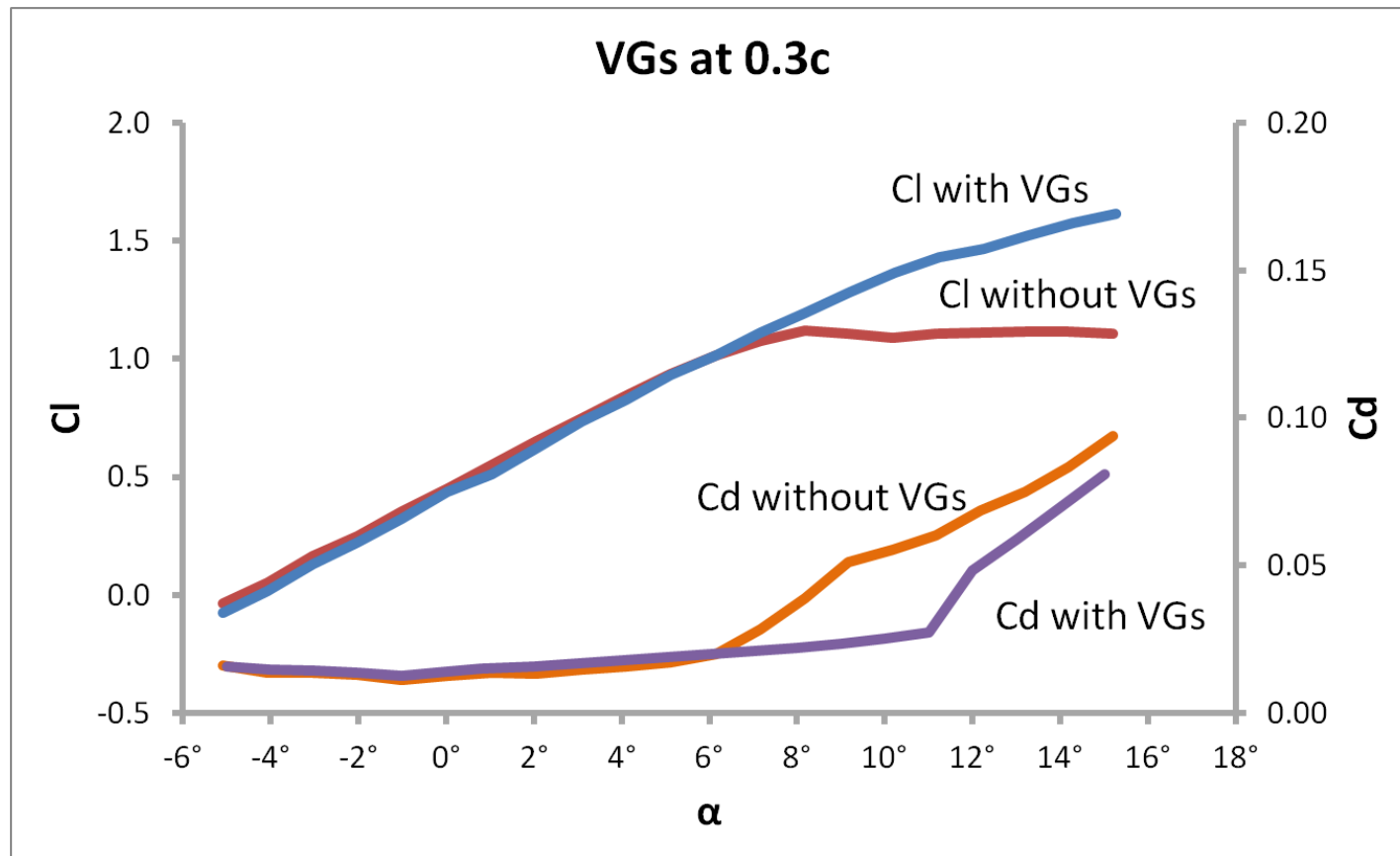


# Stall Cell Control



## Passive Vortex Generators can delay SC formation

- $C_{lmax}$  increased substantially (~50%)
- Drag penalty at lower angles of attack (~0.002)

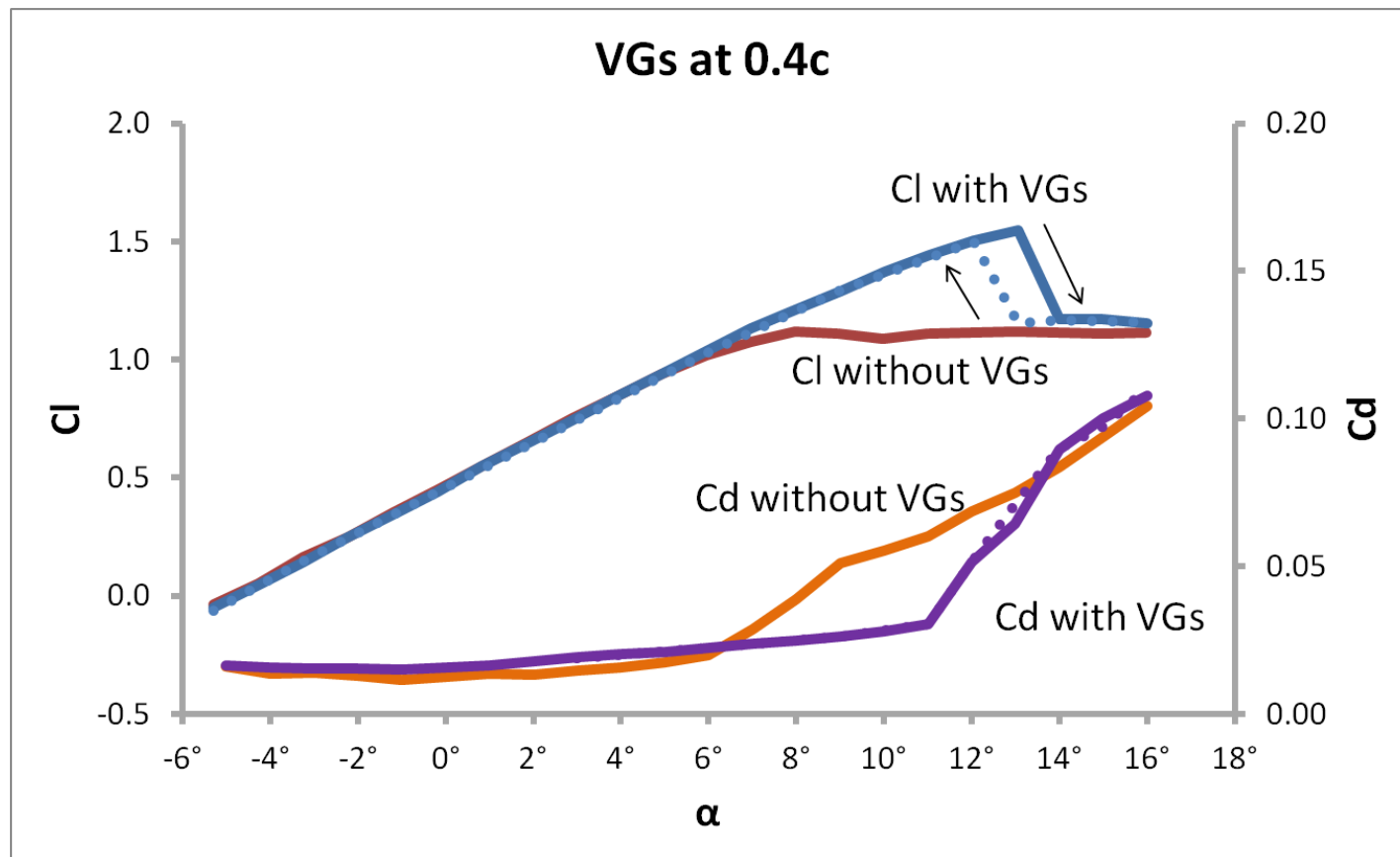


# Stall Cell Control



## Passive Vortex Generators can delay SC formation

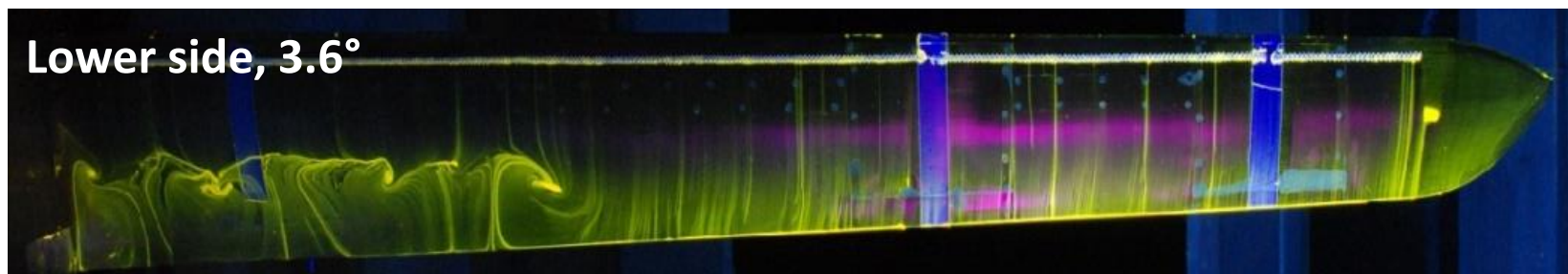
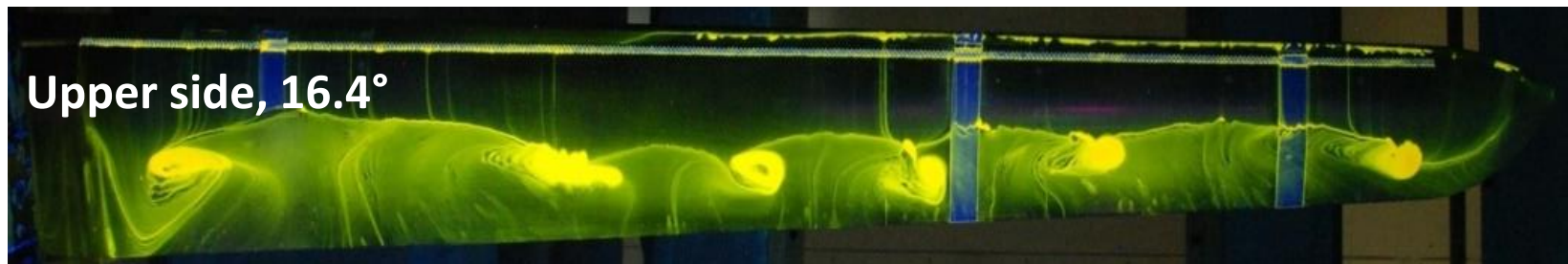
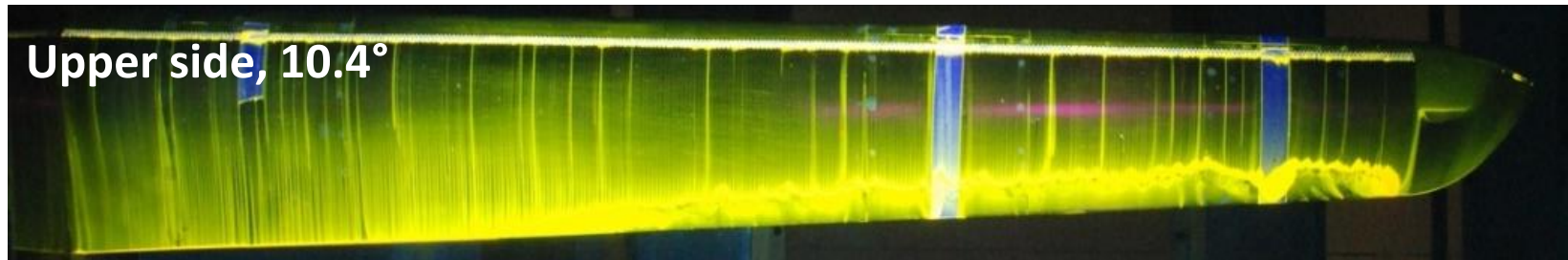
- Depending on the chordwise location of the VGs, flow hysteresis may appear



# Stall Cells, where else?



## On wind turbine blades at standstill

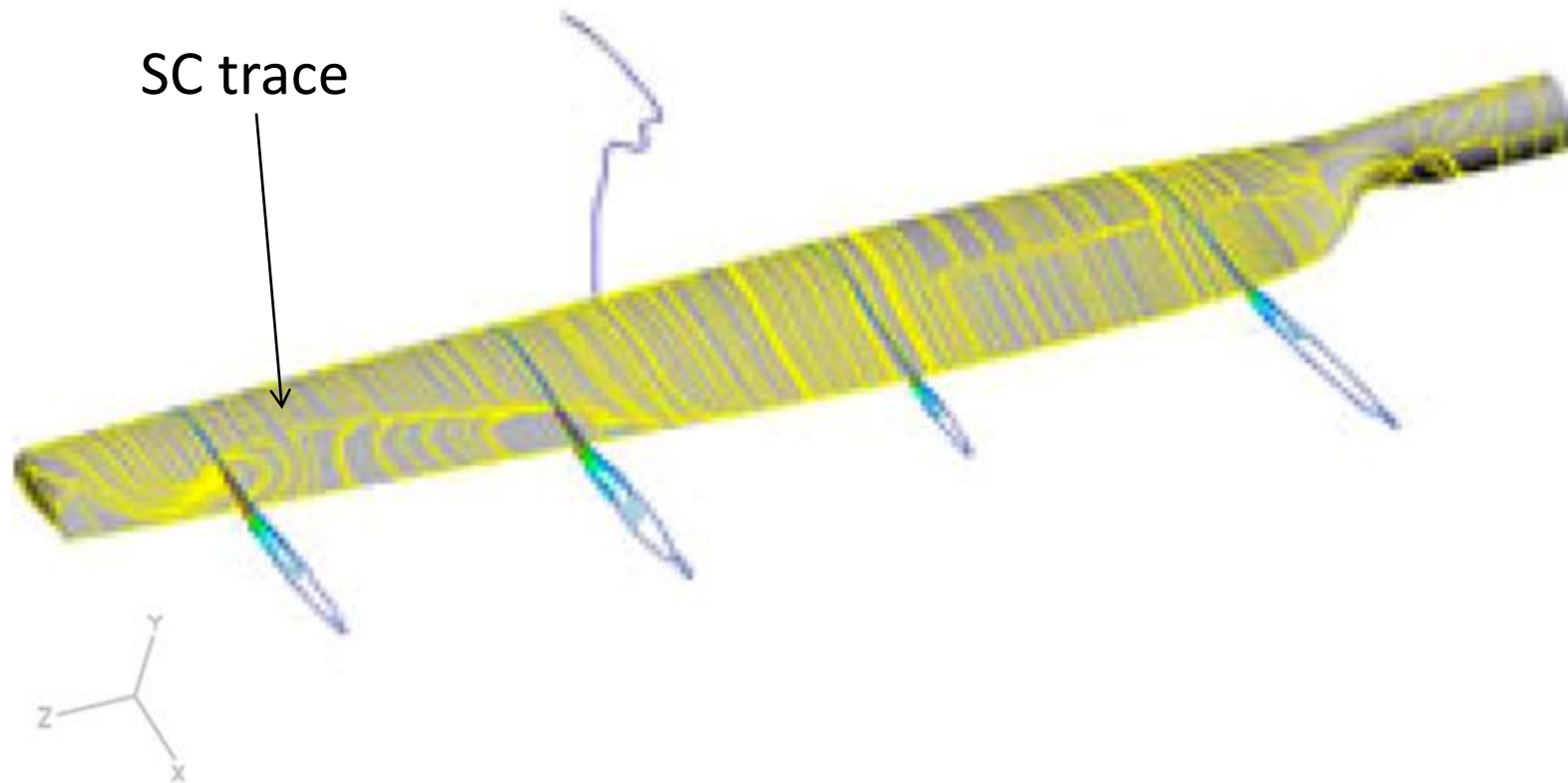


[Boorsma, 2014]

# Stall Cells, where else?



## On wind turbine blades at standstill

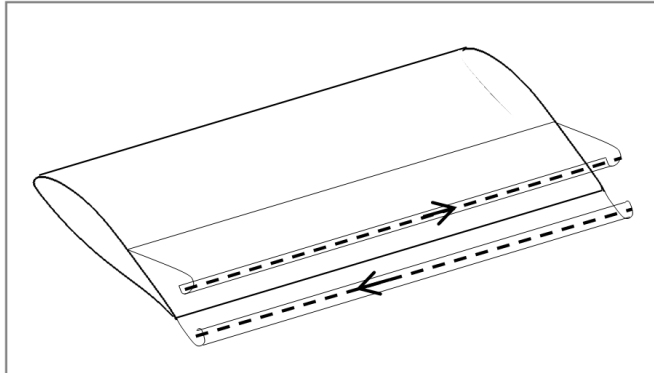


[Sorensen & Schreck, 2012]

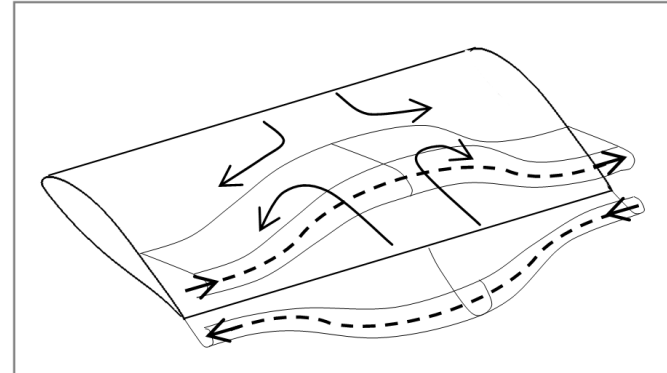
# Stall Cell Formation



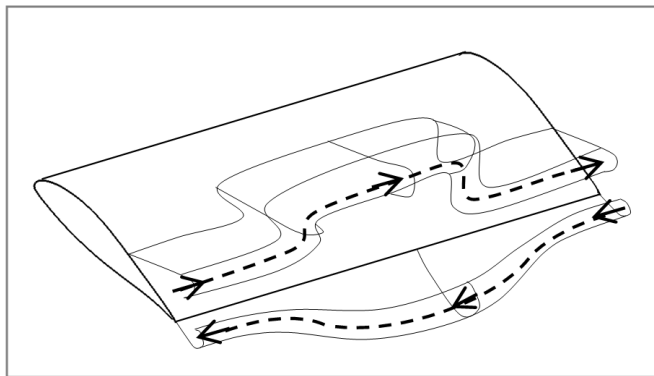
## Formation Hypothesis



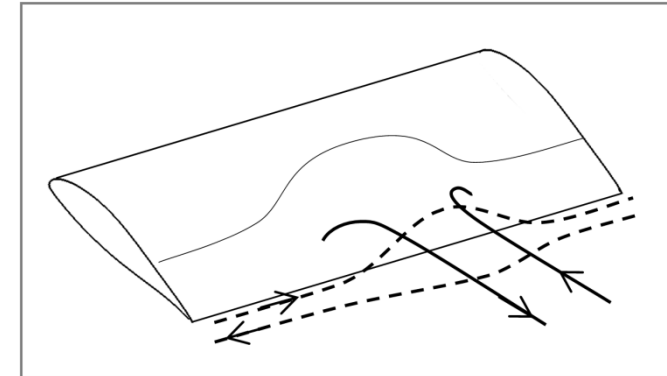
Initial 2D formation



3D instability develops



Separation shear layer folds up



Final time averaged flow

# Conclusion

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Does a symmetric 2D set-up result in symmetric 2D flow?

**Yes, under attached flow conditions**

**Not in the case of 3D trailing edge separation**

# Outlook

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What causes the Stall Cell unsteadiness?

How do adjacent Stall Cells interact with each other?

What is the link between Stall Cells and blade vibrations?

What happens on a rotating blade?

How do we determine whether a numerical solution is “real” or not?



Thank you for your attention.

Questions?



# References

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